

# A multiple-impact origin for the Moon

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The hypothesis of lunar origin by a single giant impact can explain some aspects of the Earth-Moon system. However, it is difficult to reconcile giant-impact models with the compositional similarity of the Earth and Moon without violating angular momentum constraints. Furthermore, successful giant-impact scenarios require very specific conditions such that they have a low probability of occurring. Here we present numerical simulations suggesting that the Moon could instead be the product of a succession of a variety of smaller collisions. In this scenario, each collision forms a debris disk around the proto-Earth that then accretes to form a moonlet. The moonlets tidally advance outward, and may coalesce to form the Moon. We find that sub-lunar moonlets are a common result of impacts expected onto the proto-Earth in the early Solar System and find that the planetary rotation is limited by impact angular momentum drain. We conclude that, assuming efficient merger of moonlets, a multiple-impact scenario can account for the formation of the Earth-Moon system with its present properties.

he Moon's origin remains enigmatic. The leading theory for the Moon's formation posits a scenario in which a Marssized planetesimal impacts the late-stage accreting Earth. The ejected material produces an Earth-orbiting disk, which later gravitationally accretes to a single Moon. Impact simulations found that the projectile contributes more than 70% to the disk's mass¹. This skewed mass contribution is a well-recognized problem as more high-precision measurements of isotopes indicate that the Moon and Earth are isotopically similar in oxygen ( $^{17}\text{O}/^{16}\text{O}$  and  $^{18}\text{O}/^{16}\text{O}$  within  $12\pm3$  ppm; ref. 2), titanium ( $^{50}\text{Ti}/^{47}\text{Ti}$  within  $\pm4$  ppm; ref. 3) and pre-late veneer tungsten ( $^{182}\text{W}/^{184}\text{W}$ ; ref. 4). Isotopic equilibration with a hot protoplanetary atmosphere is efficient for oxygen but insufficient to explain the similarity in more refractory elements such as titanium³.5.

Deriving more disk material from the proto-Earth occurs in impact scenarios with increased angular momentum<sup>6,7</sup> beyond the present value, which is later dissipated by an orbital resonance or an associated limit cycle<sup>8</sup>. Studies of planetary accretion<sup>9,10</sup> have shown that the equal-sized impactors are extremely rare unless assuming a very early event, which is inconsistent with the recent Moon formation timing estimates<sup>11</sup>. The expected impacts include Moonto Mars-sized impactors, supporting the specific suggested scenario of a fast-spinning Earth<sup>7</sup>; however, the predicted velocity and impact parameters' phase space is much wider than the preferred range (fast and low-angle impactors). Alternatively, *N*-body simulations have been used to argue that Earth impactors are likely to be compositionally similar to Earth, relative to other bodies<sup>12</sup>.

In this paper, we consider a multi-impact hypothesis for the Moon's formation  $^{13,14}$ . In this scenario, the proto-Earth experiences a sequence of collisions by medium- to large-size bodies  $(0.01-0.1M_{\oplus})$  (Fig. 1a,d). Small satellites form from the impact-generated disks (Fig. 1b,e) and migrate outward controlled by tidal interactions, faster at first, and slower as the body retreats away from the proto-Earth (Fig. 1c). The slowing migration causes the satellites to enter their mutual Hill radii and eventually coalesce to form the final Moon  $^{15}$  (Fig. 1f). In this fashion, the Moon forms as a consequence of a variety of multiple impacts in contrast to a more precisely tuned single impact. A similar scenario using smaller  $(0.001-0.01M_{\oplus})$ , high-velocity, late-accreting impactors was previously suggested  $^{14}$ , but supporting calculations were not provided.

N-body simulations of terrestrial planet accretion <sup>16</sup> show that the final angular momentum of the Earth's system is a result of several impacts. The largest impactor is not necessary the last one. That study shows that the majority of single collisions with Earth cannot form the present Moon because the impact angular momentum is insufficient relative to that of the current value,  $L_{\rm EM}$ . Similar to the angular momentum of the Earth–Moon system, we argue that the Moon's mass is also the result of contributions from several last impactors.

In the multi-impact scenario, the 'compositional crisis' described above is mitigated by two effects. First, since the mass and angular momentum of the present Earth–Moon system provide constraints on the sum of multiple impacts, rather than a single impact, the additional freedom in impact geometries enables mining more material from Earth than in the conventional scenario. Second, the oxygen signature distribution of the cumulative sum of multiple moonlets will have a reduced variance, increasing the probability of the Earth–Moon similarity compared with that from a single event.

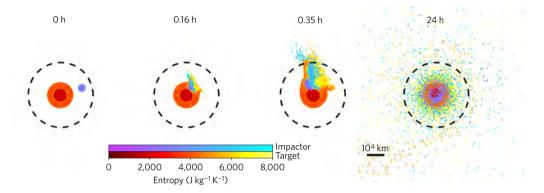
We investigate the formation of moons by multiple impacts, and consider whether Earth's Moon specifically may have been constructed by such a mechanism. As the Moon in this scenario is constructed in parts, we can sample a large phase space of initial conditions. We choose parameters for the impactor mass ratio  $\gamma$ , speed  $V_{imp}$ , direction angle  $\beta$  (relative to the line connecting the centres at contact), and planetary rotation  $\omega$  (see Supplementary Table 1 for exact values), which resemble the characteristics of the last impactors onto Earth<sup>16</sup>. High-velocity impactors are more frequent for smaller planetesimals because their eccentricities and inclinations are strongly affected by scattering events and distant interactions. In comparison with the classic impact scenario<sup>1</sup>, we consider smaller masses and a larger range of angles (from headon to near-grazing impacts). In comparison with the preferred scenario in the fast-spinning Earth scenario<sup>7</sup>, we include higher impact velocities and more slowly rotating planets.

#### Debris disk

The mass of the impact-generated debris disks is evaluated relative to the present Moon's. Of the 864 simulations performed, 750 simulations result in a disk with a discernible mass at our resolution

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**Figure 1** | **Lunar formation in the multiple-impact scenario. a,b**, Moon- to Mars-sized bodies impact the proto-Earth (**a**) forming a debris disk (**b**). **c**, Due to tidal interaction, accreted moonlets migrate outward. **d,e**, Moonlets reach distant orbits before the next collision (**d**) and the subsequent debris disk generation (**e**). As the moonlet-proto-Earth distance grows, the tidal acceleration slows and moonlets enter their mutual Hill radii. **f**, The moonlet interactions can eventually lead to moonlet loss or merger. The timescale between these stages is estimated from previous works<sup>1,27,32</sup>.



**Figure 2** | Impact simulation. Several snapshots of one of the simulations with initial conditions of  $\gamma = 0.025$ ,  $V_{imp} = 2V_{esc}$ ,  $\beta = 30^{\circ}$  and  $\omega = 0.5\omega_{max}$ . The colour bars represent the entropy of the impactor and target. All projections are on the equatorial plane with one hemisphere removed. The impactor core is shown over the target. The Roche limit is represented by the dashed line.

 $(>10^{-3}M_{\rm moon})$ . The most massive disk found is  $1.20M_{\rm moon}$ , which results from a graze-and-merge impact with the highest mass impactor at a speed of  $\sim 1V_{\rm esc}$ , onto the most rapidly rotating target  $(\omega=0.5\omega_{\rm max})$ , where  $\omega_{\rm max}$  is the planetary rotation rate at tidal breakup) within our parameter space.

An off-axis energetic impact onto a rotating target is found to produce a debris disk that is sourced from both impactor and target material. The cores are seen to merge, and the heating is concentrated in the upper planetary mantle (Fig. 2). The high-angle impacts ( $\beta=45^\circ$ ) almost always produce a disk because the angular momentum of the impactor, even for relatively low velocities near  $V_{\rm esc}$ , is sufficient to impart enough angular momentum to the ejecta to form a disk (Fig. 3). The head-on and low-angle impacts require a high velocity to eject material to orbit. Moreover, high-angle (grazing) impacts result in little mixing between the planet and the disk with the impactor contributing a substantial fraction to the disk. Low-angle impacts produce disks with a higher degree of compositional similarity to Earth.

We find that the composition of the disk does not strongly depend on the retrograde sense ( $L_{\rm imp} < 0$ , where the impactor's angular momentum is opposite to that of the planet) or prograde sense ( $L_{\rm imp} > 0$ ) of the impact.

Several trends in disk mass emerge. The majority of initial conditions within the wide phase space we tested result in disks that yield a high compositional difference,  $|\delta f_{\rm T}| > 10\%$ . However, the fast impactors produce disks with a low value of  $\delta f_{\rm T}$  and a lack of iron. Significant fractions of the mass of fast, high-angle impactors escape the system, leaving material originating from the planet's mantle in orbit.

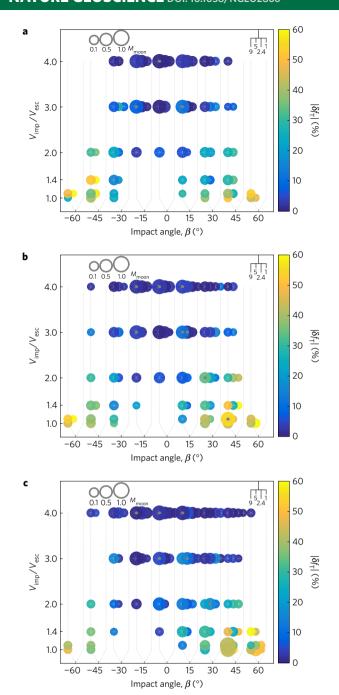
In addition to these trends, we also find abrupt transitions in disk mass. At high impact angles and low velocities, disk masses span a wide range varying by as much as a factor of ten. These scenarios transition from graze and merge to partial accretion (previously defined<sup>17</sup>) as the mass ratio increases. The graze-and-merge

scenario exhibits a qualitatively different character, as the impactor collides with the target a second time resulting in significantly enhanced mass ejection.

The mass of the satellite increases with impact angular momentum in the partial-accretion regime (Fig. 4a). Again, the maximum satellite mass is achieved by impacts that graze and merge. Satellites formed by hit-and-run collision show large relative mass variations but they are all smaller than  $\sim 0.2 M_{\rm Moon}$ .

For head-on impacts only planets with initial rotation produce a moonlet because the angular momentum required for the disk to remain stable against collapse must originate from the planetary rotation (Fig. 4b). Head-on impacts with slow rotation produce disks with a low contribution of material from the impactor. However, they lack angular momentum and a moonlet cannot form. Adding rotation to the planet increases the angular momentum of the disks, rendering the initial rotation of the planet crucial in producing moonlets by low-angle impacts. The energetic hitand-run regimes are easily distinguishable as the final angular momentum is close to the initial angular momentum (represented by the dashed lines), independent of the impactor's angular momentum. In contrast, in the accretionary scenarios the angular momentum changes by an amount proportional to the impactor's initial angular momentum. The resulting systems sometimes exceed  $L_{\rm EM}$  although the values are lower than those found in past work<sup>6,7</sup> because the impacts here are smaller. We note that subsequent impacts in a multiple-impact scenario may further alter this value. The high-angular-momentum and high-energy cases can cross the 'hot spin stability limit'18, explaining the Moon's enrichment in heavy potassium isotopes<sup>19</sup>.

We conclude that medium impactors (0.01–0.1 $M_{\oplus}$ ) can produce a sub-lunar-size moonlet whose composition ranges from impactor dominated to target dominated. Near head-on impactors are preferred because they efficiently incorporate planetary material in the disk. Within the parameter space investigated, we find that the

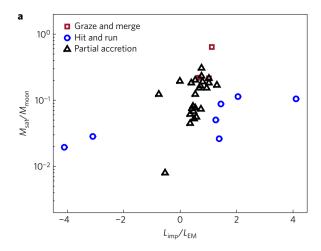


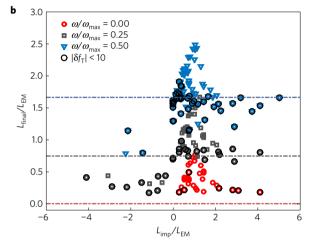
**Figure 3** | **Disk properties in the angle-velocity phase space. a-c**, Initial planetary rotation rate  $\omega=0.10\omega_{\rm max}$  (**a**),  $\omega=0.25\omega_{\rm max}$  (**b**) and  $\omega=0.50\omega_{\rm max}$  (**c**). The marker size corresponds to disk mass and the colour to the compositional difference between the silicates in the final planet and disk. For comparison, the grey circles in the upper left corner represent a disk mass of 0.1, 0.5 and 1  $M_{\rm Moon}$ . Markers are shifted horizontally according to the mass ratio, from left to right (9, 5, 2.4, 1%). The grey dots indicate disks that have an iron content larger than the estimated lunar core mass of 0.1 $M_{\rm moon}$ . Disks containing <100 smoothed particle hydrodynamics (SPH) particles were omitted.

retrograde impactors often fail to form a disk with enough angular momentum to accrete a moonlet.

### **Proto-Earth**

Post impact the planet may gain or lose mass depending on the balance of impactor merger and net erosion of material. Overall, the majority of the cases examined produce a partial or perfect merger;

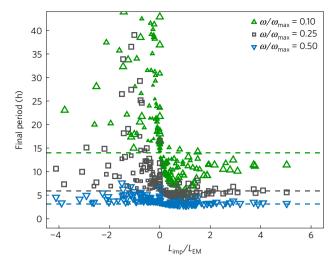




**Figure 4** | **Final satellite mass and system angular momentum. a**, Mass of the formed satellite using equation (2) as a function of impact angular momentum for  $\omega=0.25\omega_{\rm max}$  rotation. The colours correspond to different collisional regimes (hit and run—impactor escapes partially intact; graze and merge—impactor impacts the target twice; partial accretion—addition of mass to the target). **b**, The final angular momentum of all the systems that created a satellite. The different styles of markers represent different initial rotations. The darker horizontal lines represent the initial planetary angular momentum value with colours corresponding to the colours of the markers. Disks containing <100 $N_{\rm SPH}$  were omitted.

high-energy cases result in net erosion of the planet (Supplementary Fig. 3). Extreme cases ( $\gamma=0.091;\ V_{\rm imp}=4V_{\rm esc};\ \beta=0^{\circ}$  at all initial rotations) even strip half the planet's mass. These findings are in good agreement with past studies<sup>17</sup>, which found that planetary erosion is more common at lower impact angles than at high angles due to the larger interacting mass and interacting energy (defined in the Methods).

For initially slowly rotating targets, the spin accelerates (the period decreases) due to typical prograde impacts (Fig. 5). However, the rotation rate saturates for more rapidly rotating planets, which are difficult to accelerate further due to angular momentum drain carried away by ejected material. The retrograde impacts consistently decelerate the planet. The planetary angular momentum may be decreased by both material ejection and retrograde impacts. Hence, in the limit of a large number of impacts, angular momentum drain limits the otherwise random-walk growth of the angular momentum vector. As shown, for an initial rotation rate of 5.9 h ( $\omega=0.25\omega_{\rm max}$ ), retrograde collisions decelerate the planet while prograde collisions hardly change the period, so that accelerating beyond this period is difficult.



**Figure 5** | **Planetary rotation.** Post impact rotation as a function of the initial impact angular momentum. The marker size represents the mass of the impactor and the marker style and colour represent the initial rotation rate of the target, also indicated by the dashed lines for each group.

Moreover, changes in the direction of the planetary spin in one single event are seldom large. We observed three retrograde cases where the small planetary spin ( $\omega=0.1\omega_{\rm max}$ ) changed its direction. Therefore, we conclude that usually subsequent moonlets will have the same sense of rotation (both prograde or both retrograde).

## Disk structure

The disk structure, including its entropy, density and vapour fraction are important in determining the constituents and efficiency of moonlet accretion. Tidal forces within the Roche limit prevent accretion; therefore, the inner disk material must spread beyond the Roche limit before it can aggregate. The accretion of material outside the Roche limit is efficient, whereas the accretion of the inner disk is self-limiting because the newly spawned moonlets at the boundary will confine the disk and cause disk mass to fall onto Earth<sup>20</sup>. In the accretionary simulations here disks of non-negligible mass are well extended beyond the Roche limit, with more than 60% of their mass outside this radius. Although initial rotation has been shown to reduce this fraction<sup>21</sup>, we conclude here that the radial mass distribution of the disk is amenable to moonlet formation, subject to angular momentum constraints (discussed above).

The surface density is maximum inside the Roche limit, but is lower than in previously proposed scenarios (refs 1,6,7 as calculated by ref. 21) due to the lower disk mass in our scenario (Fig. 6a,c). The initial entropy of the disk will determine the amount of volatiles incorporated in the disk and the cooling time required to condense the silicate atmosphere<sup>5</sup>. We find that the specific entropy is approximately constant throughout the disk, with deviations of only  $\sim$ 10% from the mean (Fig. 6b-d). The mean entropy grows with impact energy, but while the impact energy ranges over more than two orders of magnitude, the mean entropy increases by only a factor of three. This may be understood by recognizing that the shock wave that accelerates material to bound orbits is also responsible for heating the material. Thus, the addition of kinetic energy and the addition of thermal energy are linked and ejected material that remains bound in the disk cannot be heated arbitrarily. When significant erosion occurs, the disk entropy is reduced because more energy is expended in ejecting material to escaping trajectories.

## Implications for Moon formation

By releasing some of the constraints on the impact-generated debris disk mass and system angular momentum due to individual impacts, the multiple-impact scenario allows a significantly expanded range of parameter space compared with previous Moon formation scenarios<sup>1,6,7,22</sup>. Freedom in impact geometry and velocity allows mining more material from Earth, and the sum of such impact-generated moonlets may naturally lead to the current values of the Earth–Moon system. Most terrestrial rocks have similar ratios of <sup>182</sup>W/<sup>184</sup>W; however, excesses in these ratios were found for Kostomuksha komatiites rocks dated at 2.8 billion year ago<sup>23</sup>, suggesting an unmixed mantle reservoir. Evidence of an unmixed mantle was also observed in noble gas samples<sup>24</sup>. Efficient mantle mixing is predicted for the single-impact high-angular-momentum scenarios<sup>25</sup>, erasing any primordial heterogeneity that predated Moon formation. Multiple smaller impacts promote preservation of primordial heterogeneity of Earth's mantle, and potentially, also contribute to that of the Moon<sup>15,26</sup>.

We find that debris disks resulting from medium- to large-size impactors  $(0.01-0.1M_{\oplus})$  have sufficient angular momentum and mass to accrete a sub-lunar-size moonlet. We performed 1,000 Monte Carlo simulations of sequences of N = 10, 20 and 30 impacts each, to estimate the ability of multiple impacts to produce a Moon-like satellite. The impact parameters were drawn from distributions previously found in terrestrial formation dynamical studies<sup>27</sup>. With perfect accretionary mergers, approximately half the simulations result in a moon mass that grows to its present value after  $\sim$ 20 impacts (see Supplementary Methods). We verified that the standard deviation of the  $\Delta^{17}$ O difference between the growing moon and target is reduced initially as  $\sqrt{N}$ , as expected. After ~20 impacts (roughly equivalent to the target mass) the compositional difference stabilizes, with each growing body maintaining a signature, owing to different proportions contributed by the incoming sources. In the canonical giantimpact scenario, only  $\sim$ 1-2% of impactors lead to the observed compositional similarity (assuming the same impactor composition distribution as described, and that the impactor constitutes 60-80% of the disk<sup>28,29</sup>), while in multiple-impact scenarios the fraction of successful simulations increases to tens of per cent, with the precise values depending on the assumed distribution. The range of conditions producing a Moon-mass satellite with terrestrial composition is confined to a narrow set of conditions of fast  $(V/V_{\rm esc} \sim 4)$ , near-head-on events ( $\beta \sim 0$ ). If multiple impacts are considered, additional components with lower impact velocity are allowed, such as  $V/V_{\rm esc} \sim 2$  with  $|\beta| < 30^{\circ}$ . This occurs to some extent because of the reduction of the variance composition in averaging multiple components, but also importantly, because there are regions in phase space where the predicted moonlet masses are small and the compositions are Earth-like. Such sub-lunar components can still provide the building blocks for the final Moon, while in the single-impact hypothesis they are rejected due to their small mass.

Preimpact planetary rotation promotes moonlet formation by facilitating the formation of a more massive disk from more loosely bound planetary material, and by increasing the angular momentum of the disk, stabilizing it against collapse. But the preimpact rotation rate is limited. Our hydrodynamic simulations show that multiple impactors with isotropic directions cannot accelerate the planet to breakup rotation due to angular momentum drain by escaping material. The rotation of an Earth-like target begins to saturate at a period  $\sim$ 5.9 h, in contrast to previous calculations that assume a perfect merger. Faster rotation rates may be conceivably achieved by rare single large events, but they are not expected to develop from the sum of multiple collisions. This is validated by the Monte Carlo simulations of impact sequences showing the vast majority of cases remain within the range  $-0.5\omega_{\rm max} \le \omega \le 0.5\omega_{\rm max}$ (Supplementary Fig. 4c). For sequences of 10, 20 and 30 impacts, the planetary spin migrated outside this range for only 1, 7 and 13% of the cases, respectively. Most cases have a final rotation rate  $-0.2\omega_{\text{max}} \le \omega \le 0.2\omega_{\text{max}}$ . This neglects the change in planetary

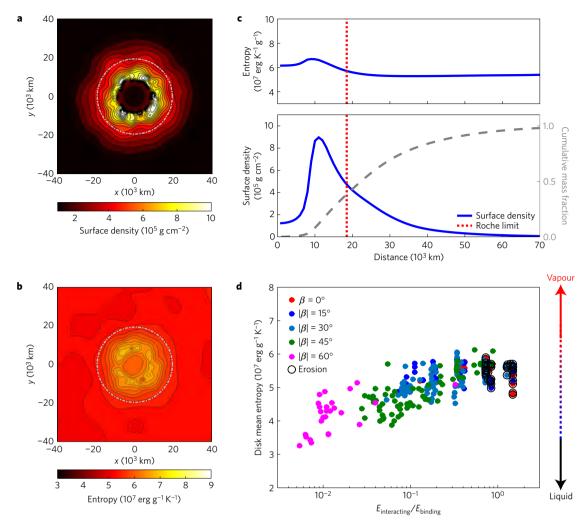


Figure 6 | Disk structure post impact. a,b, Two-dimensional structure of the surface density (a) and entropy (b) of the disk. c, Zonal average of the surface density and entropy after circularization of the disk versus the distance from the planet. The white dashed lines in a and b, and the red dashed lines in c represent the Roche limit. The initial conditions are the same as in Fig. 2. d, For our complete set of simulations, mean entropy depends on the energy of the interacting mass (see Methods), normalized by the gravitational biding energy of the target ( $E_{binding} = GM_{target}^2/R_{target}$ ). Cases exhibiting planetary erosion (defined as  $M_{planet} \le 0.85M_{target}$ ) are indicated by black circles.

rotation due to moonlet tidal evolution, which would generally decrease the spin rate.

The preferred scenario of the single-impact high-angular-momentum case<sup>7</sup> invokes a small retrograde impactor that produces a massive satellite due to the rapid initial rotation assumed for the planet (near the breakup rate). Our conclusions, on the limited planetary rotation gain and the resulting debris disk masses, point to typical individual moonlet masses being considerably below the present Moon mass. Specific scenarios that exceed one lunar mass may be constructed. However, the broad range we find within the typical parameter space that leads to sub-lunar mass disks supports the notion that the Earth's Moon was formed by the merger of multiple moonlets.

Satellite pairs formed by the same impact were found to be mostly unstable  $^{30}$ , leading to moonlet–Earth collisions, moonlet–moonlet mergers or scattering. A high percentage of moonlet merger was found for cases in which the inner moonlet is larger, and hence tidally evolves faster than the outer moonlet. The multiple-impact scenario operates on longer timescales, allowing even a smaller inner moonlet to reach the Hill radius of the outer moonlet in suitable time ( $\sim 100 \, \mathrm{Myr}$  for a  $0.1 M_{\mathrm{moon}}$  inner moonlet and a  $0.9 M_{\mathrm{moon}}$  outer moonlet at  $30 R_{\mathrm{Earth}}$ , assuming current tidal values). Moreover, the survival of an accreted moonlet depends also on

the planetary environment, as collisionless encounters of leftover planetesimals can excite the satellite's eccentricity and possible loss<sup>31</sup>. Future work will address the new dynamics and merger efficiency of the accreted moonlets to a final Moon.

#### Methods

Methods, including statements of data availability and any associated accession codes and references, are available in the online version of this paper.

Received 14 September 2016; accepted 30 November 2016; published online 9 January 2017

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#### **Acknowledgements**

We thank S. Stewart and R. Citron for providing guidance on the computational code, as well as A. Mastrobuono-Battisti for providing the data used for the Monte Carlo simulations. This project was supported by the Minerva Center for Life Under Extreme Planetary Conditions as well as by the I-CORE Program of the PBC and ISF (Center No. 1829/12). R.R. is grateful to the Israel Ministry of Science, Technology and Space for their Shulamit Aloni fellowship. H.B.P. also acknowledges support from the Israel-US bi-national science foundation, BSF grant number 2012384, and the European union career integration grant 'GRAND'.

#### **Author contributions**

R.R. performed the SPH simulations and their analysis with guidance by O.A. H.B.P. suggested the multiple-impact idea. All authors contributed to discussions, interpretations and writing.

## **Additional information**

Supplementary information is available in the online version of the paper. Reprints and permissions information is available online at www.nature.com/reprints. Correspondence and requests for materials should be addressed to R.R.

## **Competing financial interests**

The authors declare no competing financial interests.

NATURE GEOSCIENCE DOI: 10.1038/NGE02866 ARTICLES

#### Methods

We probe a broad phase space by simulating impact scenarios with ranging values of the impactor's velocity, mass, angle and initial rotation of the target. We use smoothed particle hydrodynamics (SPH) to simulate impacts in the gravity-dominated regime, using the astrophysical code Gadget2 (ref. 33). In this implementation, the individual evolution of spherically symmetric particles is tracked in three dimensions. The spatial distribution of each particle is defined by a spline density weighting function, known as the kernel, and a characteristic radius, known as the smoothing length. The functional form of the kernel does not change during the simulation, but the smoothing length of each particle is varied to maintain a constant desired number of overlapping particles, or neighbours. This procedure allows low-density regions to be resolved, although spatial resolution is reduced. The kinematic state of each particle is evolved due to gravity, compressional heating/expansional cooling, and shock dissipation. Material strength and fracture are neglected, and for the simulated time of the impact (∼1 day) radiative processes may be ignored. A standard prescription of artificial viscosity is used to mimic shock dissipation.

We employ a tabulated equation of state that was generated using the well-tested semi-analytical equation of state M-ANEOS<sup>34</sup> with forsterite comprising the mantle and iron comprising the core.

We performed over 800 simulations consisting of  $10^5$  SPH particles to span the parameter space and a smaller number of high-resolution simulations with  $10^6$  particles to better define the characteristics of the disk.

For ease of comparison with previous work<sup>7,28</sup>, all impacts are assumed to occur in the equatorial plane of the target. The runs cover 24 h of simulation time after impact. To test consistency of our hydrodynamic code and analysis algorithms, we reproduced previously published scenarios<sup>7</sup> and obtained small differences (for example, <5% in disk mass).

We use a target with a mass of  $\sim$ 1 Earth mass because we assume that the Moon-forming impacts occur near the final stages of the planet's accretion when the target mass changes by a relatively small amount with each impact. In off-axis impacts, the fraction of the impactor mass whose projected initial trajectory intersects the planet is referred to as the interacting mass,  $M_{\rm interacting}$ , with kinetic energy  $E_{\rm interacting}$ .

We validated our assumption that previous accreted moonlets do not change the initial outcome of the disk, by simulating impacts on targets with an orbiting moon. We chose a Mars-sized orbiting body with a range of semimajor axes, from the closest stable orbit in the SPH simulation to the current location of the Moon  $((1.2-20)R_{\rm Roche})$  and an equal-sized impactor. The results show that a pre-existing moonlet does not affect the disk (mass and composition) as long as it is not close to the Roche limit ( $>3R_{\rm Roche}$ ). For very close-in satellites, the impact can induce collision in the planet—moon system. For very distant satellites, the impact can excite the moons into escaping trajectories. Massive moons, such as the one tested, will migrate quickly after their formation due to tidal dissipation ( $\sim$ O(3) yr from  $R_{\rm Roche}$  to  $3R_{\rm Roche}$  for current dissipation parameters); hence, pre-existing moons will be distant enough before the next impactor so that we may restrict our attention to simulating only two bodies. We note that, we test only the first  $\sim$ 24 h after the impact, while the longer disk evolution can be affected by resonances and angular momentum transfer with pre-existing moons  $^{35}$ .

**Impact characteristics.** We characterize the impact event by determining the mass of the disk produced, mass of the planet and angular momentum and composition of the components. After each simulation we follow an iterative procedure  $^{6.36}$  to classify particles into one of three categories: planet particles whose angular momentum is insufficient to escape the gravitational pull of the planet; disk particles whose orbital pericentre is outside the planetary radius; and escaping particles, which are gravitationally unbound. We chose to classify disk particles as those with periapsis distance larger than the equatorial radius of the planet, as opposed to using the semimajor axis equivalent distance  $(a_{\rm eq}=l_z^2/{\rm GM_{planet}})$ , because the latter overestimates the disk mass in cases where the disk particles are not vaporized, eccentric and not likely to further gravitationally interact. (We note that in high-energy and high-angular-momentum cases  $^{18}$ , this distinction is irrelevant). In this classification we neglect pressure gradients as particles are assumed to travel on a Keplerian orbit, as would solid particles.

The compositional difference between the silicates in the final planet and disk is defined as  $^{6.7}$ :

$$\delta f_{\rm T} = \left[ f_{\rm disk,tar} / f_{\rm planet,tar} - 1 \right] \times 100 \tag{1}$$

where  $f_{\rm disk,tar}$  and  $f_{\rm planet,tar}$  are the mass fractions of silicate originating from the target in the disk and the final planet, respectively. A value of  $\delta f_T < 0$  implies that the disk contains more material derived from the impactor than from the target. If the impactors had the same isotopic composition as Mars then the compositional constraint allows only  $|\delta f_T| < 2$  (ref. 5); however, the projectile may have been more

similar to Earth than to Mars<sup>12</sup>. The sum of multiple disks will contribute to the final moon's composition, so each disk can vary to some extent while maintaining an average of small  $|\delta f_{\rm T}|$ .

The slower accretion of the moonlet from the disk is not simulated but estimated from the calculated disk mass,  $M_{\rm disk}$ , and its angular momentum,  $L_{\rm disk}$ , using results from previous N-body simulations<sup>32</sup>:

$$M_{\rm sat} = 1.9 \frac{L_{\rm disk}}{\sqrt{\rm GM_{planet}} a_{\rm R}} - 1.15 M_{\rm disk} - 1.9 M_{\rm esc}$$
 (2)

where G is the gravitational constant. This relation is an estimate for the satellite mass as the exact accretion efficiency is still uncertain. We assume that the escaped mass during the accretion process is negligible ( $M_{\rm esc}=0$ ) as in previous studies of the equation is not valid for disks with high specific angular momentum (yielding  $M_{\rm sat} > M_{\rm disk}$ ) that correspond to moon accretion at distances greater than  $1.3R_{\rm Roche}$ . We neglect the disks that yield a negative angular momentum compared with Earth, as a moonlet in a retrograde orbit will experience tidal acceleration that will lower the semimajor axis until it collides with the planet.

We assume that initially eccentric and inclined disks are circularized and flattened as collisions between particles in the disk will lead in a few orbital periods to a decrease in orbital eccentricities and inclinations while angular momentum is conserved. The final semimajor axis of each disk particle is  $a_{\rm eq} = l_z^2/{\rm GM_{planet}}$ , where  $l_z$  is the specific angular momentum of the particle normal to the equatorial plane of the planet. By lowering all the disk particles to the equatorial plane, assigning them a random phase and smoothing them according to the kernel function, we calculate the surface density and specific entropy of the disk at any given point.

Monte Carlo simulations. We performed Monte Carlo simulations of impact sequences by drawing impacts parameters (mass ratio,  $V_{\rm imp}/V_{\rm esc}$ , impact angle  $\beta$ ) within our phase space of interest and distributions based on previous terrestrial formation studies<sup>37</sup>. The angle distribution is symmetric, allowing retrograde and prograde impacts. Due to the uncertainty in the impact parameters, we also separately examined sequences of high-velocity (2.5 –  $4V_{\rm esc}$ ) and low-velocity (1 –  $2.5V_{\rm esc}$ ) impactors. The impactor mass ratio of interest in this work is 1–9%, the upper limit corresponding to the approximate isolation mass <sup>37</sup> (the protoplanetary mass expected to accrete by the oligarchic growth in the early planetesimal disk) of the inner Solar System. Although rare larger events are possible, our goal here is to demonstrate the feasibility of a scenario consisting of multiple smaller events. Each impactor has a distinct compositional signature drawn from a normal distribution with a standard deviation of a Mars-like composition  $^5$ .

For every simulated impact, we perform a linear interpolation within our phase space to determine post impact conditions such as planetary rotation spin, moonlet mass, iron disk content and the impactor contribution to the disk and planet. The planetary spin and signature is evolved from one impact to the next. The final Moon's characteristics are calculated assuming perfect moonlet accretion and the difference between the planet's silicate composition and that of the growing moon's is tested. By assuming that the iron fraction in the disk is preserved in the accreted moonlet, we estimate the iron fraction in the final moon. The final angular momentum of the system is obtained from the sum of the individual disk angular momenta with Earth's final value. Perfect accretion of moonlets is assumed as a simplification, as the details and probabilities of various multi-moonlet evolutionary paths are yet unknown.

 $\label{lem:code} \textbf{Code availability.} \ The \ code \ used \ to \ generate \ the \ hydrodynamic \ simulations \ can \ be \ accessed \ at \ https://www.mpa-garching.mpg.de/gadget.$ 

**Data availability.** The data that support the findings of this study are available from the corresponding author on request.

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